

# Laser Demonstration System Options

- Laser WPT from ISS to ground
  - Approximately 5 kW available input power (constrained by ISS external facilities)
  - Requires Laser WPT module, not a full WPT spacecraft
  - Requires ground receive system
  - No free-flying receive spacecraft required
  - Lowest cost demonstration option ?
  - ISS integration issues
- Laser WPT from ISS to spacecraft
  - Approximately 5 kW available input power (constrained by ISS external facilities)
  - Requires Laser WPT module, not a full WPT spacecraft
  - Requires free-flying receive spacecraft
  - ISS integration issues
- Laser WPT from space to ground
  - Power level TBD (up to 100 kW, constrained by desire for single launch)
  - Requires free-flying WPT spacecraft in LEO (orbit TBD)
  - Requires ground receive system
- Laser WPT from space to space
  - Power level TBD (up to 100 kW, constrained by single launch for both spacecraft)
  - Requires free-flying WPT spacecraft in LEO (orbit TBD)
  - Requires free-flying receive spacecraft

# Microwave Demonstration System Options

- Microwave Beam Calibration from ISS to ground
  - Approximately 5 kW available input power (constrained by ISS external facilities)
  - Does NOT provide useful power, only beam calibration
  - Requires Microwave WPT module, not a full WPT spacecraft
  - Requires ground receive system
  - No free-flying receive spacecraft required
  - Lowest cost Microwave demonstration option ?
  - ISS integration issues
- Low-Power Microwave WPT from ISS to spacecraft
  - Approximately 5 kW available input power (constrained by ISS external facilities)
  - Requires Microwave WPT module, not a full WPT spacecraft
  - Requires free-flying receive spacecraft
  - ISS integration issues
- Microwave WPT from space to ground
  - Power level TBD (up to 100 kW, constrained by desire for single launch)
  - Requires free-flying WPT spacecraft in LEO
  - Requires Ground receive system
- High-Power Microwave WPT from space to space
  - Power level TBD (up to 100 kW, constrained by single launch for both spacecraft)
  - Requires free-flying WPT spacecraft in LEO (orbit TBD)
  - Requires free-flying receive spacecraft

# Additional Demonstrator Mission Options

- Use WPT power to perform experiments on receive spacecraft
- Use array power to perform experiments on WPT spacecraft
  - High-power communications
  - Space-based radar
  - Manufacturing
  - Propellant production
  - Space science
  - Others
- Use WPT power from WPT spacecraft
  - Laser annealing of solar arrays
  - Provide energy and control to solar sails
  - Rover demonstration
  - Others
- Use array power or WPT power for electric propulsion transfer
  - Possible initial orbits: ISS, LEO 28.5 deg, Sun-Sync
  - Possible destinations: GEO, HEO, E-M L1, E-S L2, Moon, Mars, Van Allen Belts, Asteroids, Comets

# Technology Flight Demos

## Discussion Topics

- **Question:** In the proposed ISS power beaming experiment, why must one satellite be both the power beaming receiver and the 100 kW platform for the tech. experiments?

*Two different sats may make more sense. The ISS beaming experiment could be relatively simple, and could beam to a site on Earth. Use an existing beam expander (small enough to go through the JEM airlock) to put a 3m spot on ground, with power received on PV array. Could also demo retrodirective beam experiments.*

# Technology Flight Demos

## Discussion Topics

- **Question:** Should we only work laser experiments for the 1<sup>st</sup> demo?
- *Suggestion to have a credible laser demo for WPT, since concerns about what would be credible for a low power microwave demo*
- *Suggestion to leave both options open now, since laser impacts not fully understood. Do more work, then downselect technology.*
- *What about a combined laser/microwave experiment? Could demo both systems, but issues include efficiency losses of laser into microwave*

# Technology Flight Demos

## Discussion Topics

- **Question:** Should the ISC configuration be used for the 100 kW tech. demonstrator?

*ISC is complex and technically challenging, so a simpler bus may make more sense. Salient features of the ISC could be demonstrated in a small experiment on the tech. demonstrator.*

# Technology Flight Demos

## Discussion Topics

- **Question:** What about cooperative work with DARPA, to demo WPT to one of their sats?

*Erectorset on-orbit assembly may be a candidate. WPT transmission to Orbital Express? May need 2<sup>nd</sup> copy of satellites, due to 1 year life of Orbital Express sats.*